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without rotodome .
Part 3- Aerostatic Model Formulation

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15. Abstract

The mathematical models and data used in the off-line flight simulation of the rotodomed aircraft have been described in detail in Part 1. The issues related to the Rolls Royce Engine simulation studies have been presented in Part 2. In this document, static wind tunnel test data generated at H.A.L and considered as acceptable by BAE has been used as a reference data for verifying and formulating the aero-static model and associated data for HS-748 and AEW configurations. The resulting static model has been combined with dynamic model to arrive at an overall representative Aero model for use in flight simulation studies. This procedure is vital for unusual aircraft configurations as most of the data at design stage is available only from static wind tunnel tests.